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Climbing Flight Formulas

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List of 16 Climbing Flight Formulas

Climbing Flight

1) Centrifugal Force in Accelerated Flight

$$\text{fx } F_c = F_L + T \cdot \sin(\sigma_T) - m \cdot [g] \cdot \cos(\gamma)$$

[Open Calculator !\[\]\(a870788d6ed9b8fd294b7654a8c8526b_img.jpg\)](#)

$$\text{ex } 28.03926\text{N} = 200\text{N} + 700\text{N} \cdot \sin(0.034\text{rad}) - 20\text{kg} \cdot [g] \cdot \cos(0.062\text{rad})$$

2) Drag in Accelerated Flight

$$\text{fx } F_D = T \cdot \cos(\sigma_T) - m \cdot [g] \cdot \sin(\gamma) - m \cdot a$$

[Open Calculator !\[\]\(c50c8b7b2cc2cf9ff925edec0ee94c0d_img.jpg\)](#)

$$\text{ex } 80.04298\text{N} = 700\text{N} \cdot \cos(0.034\text{rad}) - 20\text{kg} \cdot [g] \cdot \sin(0.062\text{rad}) - 20\text{kg} \cdot 30.37\text{m/s}^2$$

3) Excess power

$$\text{fx } P_{\text{excess}} = v \cdot (T - F_D)$$

[Open Calculator !\[\]\(f60b7a900783ac3fd531bfd9c111be6d_img.jpg\)](#)

$$\text{ex } 37197.6\text{W} = 60\text{m/s} \cdot (700\text{N} - 80.04\text{N})$$

4) Excess power for given rate of climb

$$\text{fx } P_{\text{excess}} = RC \cdot W$$

[Open Calculator !\[\]\(83bbbd261710c59db0214aa27b2edc0d_img.jpg\)](#)

$$\text{ex } 37197.6\text{W} = 3.71976\text{m/s} \cdot 10000\text{N}$$


5) Flight path angle at given rate of climb

$$\text{fx } \gamma = a \sin\left(\frac{RC}{v}\right)$$

[Open Calculator !\[\]\(f507db636256ac11a5525ef93ec6b8d7_img.jpg\)](#)


$$\text{ex } 0.062036\text{rad} = a \sin\left(\frac{3.71976\text{m/s}}{60\text{m/s}}\right)$$



6) Lift in Accelerated Flight Open Calculator 


$$f_x F_L = m \cdot [g] \cdot \cos(\gamma) + m \cdot \frac{v^2}{R_{\text{curvature}}} - T \cdot \sin(\sigma_T)$$

$$ex \ 199.653N = 20kg \cdot [g] \cdot \cos(0.062rad) + 20kg \cdot \frac{(60m/s)^2}{2600m} - 700N \cdot \sin(0.034rad)$$

7) Rate of Climb Open Calculator 


$$f_x RC = v \cdot \sin(\gamma)$$

$$ex \ 3.717617m/s = 60m/s \cdot \sin(0.062rad)$$

8) Rate of Climb for given excess power Open Calculator 


$$f_x RC = \frac{P_{\text{excess}}}{W}$$

$$ex \ 3.71976m/s = \frac{37197.6W}{10000N}$$

9) Rate of Climb of Aircraft Open Calculator 

$$f_x RC = \frac{P_a - P_r}{W}$$


$$ex \ 3.7199m/s = \frac{38199W - 1000W}{10000N}$$

10) Thrust available for given excess power Open Calculator 

$$f_x T = F_D + \left(\frac{P_{\text{excess}}}{v} \right)$$

$$ex \ 700N = 80.04N + \left(\frac{37197.6W}{60m/s} \right)$$



11) Thrust in Accelerated Flight 

$$f_x \quad T = (\sec(\sigma_T)) \cdot (F_D + (m \cdot [g] \cdot \sin(\gamma)) + (m \cdot a))$$

Open Calculator 

ex

$$699.997\text{N} = (\sec(0.034\text{rad})) \cdot (80.04\text{N} + (20\text{kg} \cdot [g] \cdot \sin(0.062\text{rad})) + (20\text{kg} \cdot 30.37\text{m/s}^2))$$


12) Total Drag for given Excess Power 

$$f_x \quad F_D = T - \left(\frac{P_{\text{excess}}}{v} \right)$$

Open Calculator 

ex

$$80.04\text{N} = 700\text{N} - \left(\frac{37197.6\text{W}}{60\text{m/s}} \right)$$


13) Velocity in Accelerated Flight 

$$f_x \quad v = \left(\frac{R_{\text{curvature}}}{m} \cdot (F_L + T \cdot \sin(\sigma_T) - m \cdot [g] \cdot \cos(\gamma)) \right)^{\frac{1}{2}}$$

Open Calculator 

ex

$$60.3747\text{m/s} = \left(\frac{2600\text{m}}{20\text{kg}} \cdot (200\text{N} + 700\text{N} \cdot \sin(0.034\text{rad}) - 20\text{kg} \cdot [g] \cdot \cos(0.062\text{rad})) \right)^{\frac{1}{2}}$$

14) Velocity of aircraft at given rate of climb 


$$f_x \quad v = \frac{RC}{\sin(\gamma)}$$

Open Calculator 

ex


$$60.03458\text{m/s} = \frac{3.71976\text{m/s}}{\sin(0.062\text{rad})}$$



15) Velocity of Aircraft for given Excess Power [Open Calculator](#) 

$$\text{fx } v = \frac{P_{\text{excess}}}{T - F_D}$$

$$\text{ex } 60\text{m/s} = \frac{37197.6\text{W}}{700\text{N} - 80.04\text{N}}$$

16) Weight of Aircraft for given Excess Power [Open Calculator](#) 

$$\text{fx } W = \frac{P_{\text{excess}}}{RC}$$

$$\text{ex } 10000\text{N} = \frac{37197.6\text{W}}{3.71976\text{m/s}}$$










Variables Used

- **a** Acceleration (Meter per Square Second)
- **F_c** Centrifugal Force (Newton)
- **F_D** Drag Force (Newton)
- **F_L** Lift Force (Newton)
- **m** Mass of Aircraft (Kilogram)
- **P_a** Power Available (Watt)
- **P_{excess}** Excess Power (Watt)
- **P_r** Power Required (Watt)
- **R_{curvature}** Radius of Curvature (Meter)
- **RC** Rate of Climb (Meter per Second)
- **T** Thrust (Newton)
- **v** Velocity (Meter per Second)
- **W** Aircraft Weight (Newton)
- **γ** Flight Path Angle (Radian)
- **σ_T** Thrust Angle (Radian)




Constants, Functions, Measurements used

- **Constant:** **[g]**, 9.80665
Gravitational acceleration on Earth
- **Function:** **asin**, asin(Number)
The inverse sine function, is a trigonometric function that takes a ratio of two sides of a right triangle and outputs the angle opposite the side with the given ratio.
- **Function:** **cos**, cos(Angle)
Cosine of an angle is the ratio of the side adjacent to the angle to the hypotenuse of the triangle.
- **Function:** **sec**, sec(Angle)
Secant is a trigonometric function that is defined ratio of the hypotenuse to the shorter side adjacent to an acute angle (in a right-angled triangle); the reciprocal of a cosine.
- **Function:** **sin**, sin(Angle)
Sine is a trigonometric function that describes the ratio of the length of the opposite side of a right triangle to the length of the hypotenuse.
- **Measurement:** **Length** in Meter (m)
Length Unit Conversion 
- **Measurement:** **Weight** in Kilogram (kg)
Weight Unit Conversion 
- **Measurement:** **Speed** in Meter per Second (m/s)
Speed Unit Conversion 
- **Measurement:** **Acceleration** in Meter per Square Second (m/s²)
Acceleration Unit Conversion 
- **Measurement:** **Power** in Watt (W)
Power Unit Conversion 
- **Measurement:** **Force** in Newton (N)
Force Unit Conversion 
- **Measurement:** **Angle** in Radian (rad)
Angle Unit Conversion 



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