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## Lift and Circulation Formulas

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## List of 16 Lift and Circulation Formulas

## Lift and Circulation

1) Angle of Attack for Circulation developed on Airfoil
$f \mathrm{fx} \alpha=a \sin \left(\frac{\Gamma}{\pi \cdot \mathrm{U} \cdot \mathrm{C}}\right)$
Open Calculator
ex $6.506912^{\circ}=a \sin \left(\frac{62 \mathrm{~m}^{2} / \mathrm{s}}{\pi \cdot 81 \mathrm{~m} / \mathrm{s} \cdot 2.15 \mathrm{~m}}\right)$
2) Angle of Attack for Lift Coefficient on Airfoil
$f \mathrm{x} \alpha=a \sin \left(\frac{\mathrm{C}_{\mathrm{L}} \text { airfoil }}{2 \cdot \pi}\right)$
Open Calculator
ex $6.506638^{\circ}=a \sin \left(\frac{0.712}{2 \cdot \pi}\right)$
3) Chord Length for Circulation developed on Airfoil
$f \mathrm{f} C=\frac{\Gamma}{\pi \cdot U \cdot \sin (\alpha)}$
Open Calculator
$\operatorname{ex} 2.152276 \mathrm{~m}=\frac{62 \mathrm{~m}^{2} / \mathrm{s}}{\pi \cdot 81 \mathrm{~m} / \mathrm{s} \cdot \sin \left(6.5^{\circ}\right)}$
4) Circulation developed on Airfoil
$\mathrm{f}_{\mathrm{x}} \Gamma=\pi \cdot \mathrm{U} \cdot \mathrm{C} \cdot \sin (\alpha)$
Open Calculator
ex $61.93442 \mathrm{~m}^{2} / \mathrm{s}=\pi \cdot 81 \mathrm{~m} / \mathrm{s} \cdot 2.15 \mathrm{~m} \cdot \sin \left(6.5^{\circ}\right)$
5) Circulation for Single Stagnation Point
$\mathrm{fx}_{\mathrm{x}} \Gamma_{\mathrm{c}}=4 \cdot \pi \cdot \mathrm{~V}_{\infty} \cdot \mathrm{R}$
Open Calculator
ex $243.1593 \mathrm{~m}^{2} / \mathrm{s}=4 \cdot \pi \cdot 21.5 \mathrm{~m} / \mathrm{s} \cdot 0.9 \mathrm{~m}$
6) Circulation in Location of Stagnation Points
$\mathrm{fx}_{\mathrm{x}} \Gamma_{\mathrm{c}}=-(\sin (\theta)) \cdot 4 \cdot \pi \cdot \mathrm{~V}_{\infty} \cdot \mathrm{R}$
Open Calculator
ex $243.1593 \mathrm{~m}^{2} / \mathrm{s}=-\left(\sin \left(270^{\circ}\right)\right) \cdot 4 \cdot \pi \cdot 21.5 \mathrm{~m} / \mathrm{s} \cdot 0.9 \mathrm{~m}$
7) Coefficient of Lift for Airfoil
$\mathrm{f}_{\mathrm{x}} \mathrm{C}_{\mathrm{L} \text { airfoil }}=2 \cdot \pi \cdot \sin (\alpha)$
ex $0.711277=2 \cdot \pi \cdot \sin \left(6.5^{\circ}\right)$
8) Lift coefficient for lift force in body moving on fluid
$f_{x} C_{L}=\frac{F_{L}^{\prime}}{A_{p} \cdot 0.5 \cdot \rho \cdot\left(v^{2}\right)}$
Open Calculator 〔
ex $0.944451=\frac{1100 \mathrm{~N}}{1.88 \mathrm{~m}^{2} \cdot 0.5 \cdot 1.21 \mathrm{~kg} / \mathrm{m}^{3} \cdot\left((32 \mathrm{~m} / \mathrm{s})^{2}\right)}$

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9) Lift Coefficient for Rotating Cylinder with Circulation

包 $\mathrm{C}^{\prime}=\frac{\Gamma_{\mathrm{c}}}{\mathrm{R} \cdot \mathrm{V}_{\infty}}$
Open Calculator
ex $12.55814=\frac{243 \mathrm{~m}^{2} / \mathrm{s}}{0.9 \mathrm{~m} \cdot 21.5 \mathrm{~m} / \mathrm{s}}$
10) Lift Coefficient for Rotating Cylinder with Tangential Speed
$\mathrm{fx}_{\mathrm{x}} \mathrm{C}^{\prime}=\frac{2 \cdot \pi \cdot \mathrm{v}_{\mathrm{t}}}{\mathrm{V}_{\infty}}$
Open Calculator
ex $12.56637=\frac{2 \cdot \pi \cdot 43 \mathrm{~m} / \mathrm{s}}{21.5 \mathrm{~m} / \mathrm{s}}$
11) Lift Force for Body moving in Fluid
$f \mathbf{x}\left(\mathrm{~F}_{\mathrm{L}}{ }^{\prime}\right)=\frac{\mathrm{C}_{\mathrm{L}} \cdot \mathrm{A}_{\mathrm{p}} \cdot \mathrm{M}_{\mathrm{w}} \cdot\left(\mathrm{v}^{2}\right)}{\mathrm{V}_{\mathrm{w}} \cdot 2}$
Open Calculator
ex $1098.693 \mathrm{~N}=\frac{0.94 \cdot 1.88 \mathrm{~m}^{2} \cdot 3.4 \mathrm{~kg} \cdot\left((32 \mathrm{~m} / \mathrm{s})^{2}\right)}{2.8 \mathrm{~m}^{3} \cdot 2}$
12) Lift Force for body moving in Fluid of Certain Density

Open Calculator [
$f \times F_{L}=C_{L} \cdot A_{p} \cdot \rho \cdot \frac{v^{2}}{2}$
ex $1094.816 \mathrm{~N}=0.94 \cdot 1.88 \mathrm{~m}^{2} \cdot 1.21 \mathrm{~kg} / \mathrm{m}^{3} \cdot \frac{(32 \mathrm{~m} / \mathrm{s})^{2}}{2}$
13) Lift Force on Cylinder for Circulation
$f \mathrm{x} \quad \mathrm{F}_{\mathrm{L}}=\rho \cdot \mathrm{I} \cdot \Gamma_{\mathrm{c}} \cdot \mathrm{V}_{\infty}$
Open Calculator
ex $53733.98 \mathrm{~N}=1.21 \mathrm{~kg} / \mathrm{m}^{3} \cdot 8.5 \mathrm{~m} \cdot 243 \mathrm{~m}^{2} / \mathrm{s} \cdot 21.5 \mathrm{~m} / \mathrm{s}$
14) Radius of Cylinder for Lift Coefficient in Rotating Cylinder with Circulation
$\mathrm{f} \times \mathrm{R}=\frac{\Gamma_{\mathrm{c}}}{\mathrm{C}^{\prime} \cdot \mathrm{V}_{\infty}}$
Open Calculator
ex $0.900584 \mathrm{~m}=\frac{243 \mathrm{~m}^{2} / \mathrm{s}}{12.55 \cdot 21.5 \mathrm{~m} / \mathrm{s}}$
15) Tangential Velocity of Cylinder with Lift Coefficient
$f \mathrm{x} \mathrm{v}_{\mathrm{t}}=\frac{\mathrm{C}^{\prime} \cdot \mathrm{V}_{\infty}}{2 \cdot \pi}$
Open Calculator
ex $42.94398 \mathrm{~m} / \mathrm{s}=\frac{12.55 \cdot 21.5 \mathrm{~m} / \mathrm{s}}{2 \cdot \pi}$
16) Velocity of Airfoil for Circulation developed on Airfoil
$\mathrm{fx} \mathrm{U}=\frac{\Gamma}{\pi \cdot \mathrm{C} \cdot \sin (\alpha)}$
Open Calculator
ex $81.08576 \mathrm{~m} / \mathrm{s}=\frac{62 \mathrm{~m}^{2} / \mathrm{s}}{\pi \cdot 2.15 \mathrm{~m} \cdot \sin \left(6.5^{\circ}\right)}$

## Variables Used

- $\mathbf{A}_{\mathbf{p}}$ Projected Area of Body (Square Meter)
- C Chord Length of Airfoil (Meter)
- $C_{L}$ airfoil Lift Coefficient for Airfoil
- $\mathbf{C}_{\mathrm{L}}$ Lift Coefficient for Body in Fluid
- C' Lift Coefficient for Rotating Cylinder
- $F_{\text {L }}$ Lift Force on Rotating Cylinder (Newton)
- $\mathbf{F}_{\mathrm{L}}$ 'Lift Force on Body in Fluid (Newton)
- I Length of Cylinder in Fluid Flow (Meter)
- $\mathbf{M}_{\mathbf{w}}$ Mass of Flowing Fluid (Kilogram)
- R Radius of Rotating Cylinder (Meter)
- U Velocity of Airfoil (Meter per Second)
- V Velocity of Body or Fluid (Meter per Second)
- $\mathbf{V}_{\infty}$ Freestream Velocity of Fluid (Meter per Second)
- $\mathbf{V}_{\mathbf{t}}$ Tangential Velocity of Cylinder in Fluid (Meter per Second)
- $\mathbf{V}_{\mathbf{w}}$ Volume of Flowing Fluid (Cubic Meter)
- $\boldsymbol{\alpha}$ Angle of Attack on Airfoil (Degree)
- 「 Circulation on Airfoil (Square Meter per Second)
- $\Gamma_{\mathbf{c}}$ Circulation Around Cylinder (Square Meter per Second)
- $\boldsymbol{\theta}$ Angle at Stagnation Point (Degree)
- $\boldsymbol{\rho}$ Density of Fluid Circulating (Kilogram per Cubic Meter)


## Constants, Functions, Measurements used

- Constant: pi, 3.14159265358979323846264338327950288

Archimedes' constant

- Function: asin, asin(Number)

The inverse sine function, is a trigonometric function that takes a ratio of two sides of a right triangle and outputs the angle opposite the side with the given ratio.

- Function: sin, $\sin ($ Angle)

Sine is a trigonometric function that describes the ratio of the length of the opposite side of a right triangle to the length of the hypotenuse.

- Measurement: Length in Meter (m)

Length Unit Conversion

- Measurement: Weight in Kilogram (kg)

Weight Unit Conversion

- Measurement: Volume in Cubic Meter ( $\mathrm{m}^{3}$ )

Volume Unit Conversion

- Measurement: Area in Square Meter ( $\mathrm{m}^{2}$ )

Area Unit Conversion

- Measurement: Speed in Meter per Second (m/s)

Speed Unit Conversion

- Measurement: Force in Newton (N)

Force Unit Conversion

- Measurement: Angle in Degree $\left({ }^{\circ}\right)$

Angle Unit Conversion

- Measurement: Density in Kilogram per Cubic Meter (kg/m³) Density Unit Conversion
- Measurement: Momentum Diffusivity in Square Meter per Second ( $\mathrm{m}^{2} / \mathrm{s}$ ) Momentum Diffusivity Unit Conversion


## Check other formula lists

- Lift and Circulation Formulas

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