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Turbojets Formulas

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List of 14 Turbojets Formulas

Turbojets

1) Exhaust Velocity given Gross Thrust in Turbojet

$$fx \quad V_e = \frac{T_G - (p_e - p_\infty) \cdot A_e}{m_a \cdot (1 + f)}$$

[Open Calculator !\[\]\(a870788d6ed9b8fd294b7654a8c8526b_img.jpg\)](#)

$$ex \quad 212.7201\text{m/s} = \frac{1124\text{N} - (982\text{Pa} - 101\text{Pa}) \cdot 0.0589\text{m}^2}{5\text{kg/s} \cdot (1 + 0.008)}$$

2) Exhaust Velocity given Thrust in Turbojet

$$fx \quad V_e = \frac{T - A_e \cdot (p_e - p_\infty)}{m_a \cdot (1 + f)} + V$$

[Open Calculator !\[\]\(c50c8b7b2cc2cf9ff925edec0ee94c0d_img.jpg\)](#)

$$ex \quad 212.7597\text{m/s} = \frac{469\text{N} - 0.0589\text{m}^2 \cdot (982\text{Pa} - 101\text{Pa})}{5\text{kg/s} \cdot (1 + 0.008)} + 130\text{m/s}$$

3) Flight Speed given Thrust in Turbojet

$$fx \quad V = V_e - \frac{T - A_e \cdot (p_e - p_\infty)}{m_a \cdot (1 + f)}$$

[Open Calculator !\[\]\(f60b7a900783ac3fd531bfd9c111be6d_img.jpg\)](#)

$$ex \quad 130.2403\text{m/s} = 213\text{m/s} - \frac{469\text{N} - 0.0589\text{m}^2 \cdot (982\text{Pa} - 101\text{Pa})}{5\text{kg/s} \cdot (1 + 0.008)}$$


4) Gross Thrust of Turbojet given Net Thrust

$$fx \quad T_G = T + D_{\text{ram}}$$

[Open Calculator !\[\]\(83bbbd261710c59db0214aa27b2edc0d_img.jpg\)](#)

$$ex \quad 1124\text{N} = 469\text{N} + 655\text{N}$$



5) Mass Flow Rate in Turbojet given Thrust 

$$fx \quad m_a = \frac{T - A_e \cdot (p_e - p_\infty)}{(V_e - V) \cdot (1 + f)}$$

Open Calculator 

$$ex \quad 4.985527 \text{ kg/s} = \frac{469 \text{ N} - 0.0589 \text{ m}^2 \cdot (982 \text{ Pa} - 101 \text{ Pa})}{(213 \text{ m/s} - 130 \text{ m/s}) \cdot (1 + 0.008)}$$

6) Mass Flow Rate of Exhaust Gases 

$$fx \quad m_{\text{total}} = m_a + m_f$$

Open Calculator 

$$ex \quad 5.033 \text{ kg/s} = 5 \text{ kg/s} + 0.033 \text{ kg/s}$$

7) Mass Flow Rate of Exhaust Gases given Fuel Air Ratio 

$$fx \quad m_{\text{total}} = m_a \cdot (1 + f)$$

Open Calculator 

$$ex \quad 5.04 \text{ kg/s} = 5 \text{ kg/s} \cdot (1 + 0.008)$$

8) Mass Flow Rate of Turbojet given Gross Thrust 

$$fx \quad m_a = \frac{T_G - (p_e - p_\infty) \cdot A_e}{(1 + f) \cdot V_e}$$

Open Calculator 

$$ex \quad 4.993429 \text{ kg/s} = \frac{1124 \text{ N} - (982 \text{ Pa} - 101 \text{ Pa}) \cdot 0.0589 \text{ m}^2}{(1 + 0.008) \cdot 213 \text{ m/s}}$$


9) Net Thrust of Turbojet given Gross Thrust 

$$fx \quad T = T_G - D_{\text{ram}}$$

Open Calculator 

$$ex \quad 469 \text{ N} = 1124 \text{ N} - 655 \text{ N}$$



10) Net Thrust Produced by Turbojet 

$$fx \quad T = m_a \cdot (1 + f) \cdot (V_e - V) + A_e \cdot (p_e - p_\infty)$$

[Open Calculator !\[\]\(e78f798d4ea5c530c9db49e7d26e6b95_img.jpg\)](#)

ex

$$470.2109\text{N} = 5\text{kg/s} \cdot (1 + 0.008) \cdot (213\text{m/s} - 130\text{m/s}) + 0.0589\text{m}^2 \cdot (982\text{Pa} - 101\text{Pa})$$

11) Nozzle Exit Area in Turbojet 

$$fx \quad A_e = \frac{T - m_a \cdot (1 + f) \cdot (V_e - V)}{p_e - p_\infty}$$

[Open Calculator !\[\]\(aa53ad6fea213b8b2226d3077e30533a_img.jpg\)](#)


$$ex \quad 0.057526\text{m}^2 = \frac{469\text{N} - 5\text{kg/s} \cdot (1 + 0.008) \cdot (213\text{m/s} - 130\text{m/s})}{982\text{Pa} - 101\text{Pa}}$$

12) Ram Drag of Turbojet given Gross Thrust 

$$fx \quad D_{\text{ram}} = T_G - T$$

[Open Calculator !\[\]\(626ce8ac21792b9405bfddfea8e0c96a_img.jpg\)](#)

$$ex \quad 655\text{N} = 1124\text{N} - 469\text{N}$$

13) Thermal efficiency of turbojet engine 

$$fx \quad \eta_{\text{th}} = \frac{P}{m_f \cdot Q}$$

[Open Calculator !\[\]\(c1168d6a8b365d11e842ece304635fa7_img.jpg\)](#)

$$ex \quad 0.682689 = \frac{980\text{kW}}{0.033\text{kg/s} \cdot 43500\text{kJ/kg}}$$

14) Turbojet Gross Thrust 

$$fx \quad T_G = m_a \cdot (1 + f) \cdot V_e + (p_e - p_\infty) \cdot A_e$$

[Open Calculator !\[\]\(ccd39a0dc6d5afcc151e1371f9462f58_img.jpg\)](#)

$$ex \quad 1125.411\text{N} = 5\text{kg/s} \cdot (1 + 0.008) \cdot 213\text{m/s} + (982\text{Pa} - 101\text{Pa}) \cdot 0.0589\text{m}^2$$










Variables Used

- A_e Nozzle Exit Area (Square Meter)
- D_{ram} Ram Drag of Turbojet (Newton)
- f Fuel Air Ratio
- m_a Mass Flow Rate Turbojet (Kilogram per Second)
- m_f Fuel Flow Rate (Kilogram per Second)
- m_{total} Total Mass Flow Rate Turbojet (Kilogram per Second)
- P Propulsive Power (Kilowatt)
- p_∞ Ambient Pressure (Pascal)
- p_e Nozzle Exit Pressure (Pascal)
- Q Fuel Calorific Value (Kilojoule per Kilogram)
- T Net Thrust of Turbojet (Newton)
- T_G Gross Thrust of Turbojet (Newton)
- V Flight Speed (Meter per Second)
- V_e Exit Velocity (Meter per Second)
- η_{th} Turbojet Thermal Efficiency



Constants, Functions, Measurements used

- **Measurement: Area** in Square Meter (m^2)
Area Unit Conversion 
- **Measurement: Pressure** in Pascal (Pa)
Pressure Unit Conversion 
- **Measurement: Speed** in Meter per Second (m/s)
Speed Unit Conversion 
- **Measurement: Power** in Kilowatt (kW)
Power Unit Conversion 
- **Measurement: Force** in Newton (N)
Force Unit Conversion 
- **Measurement: Mass Flow Rate** in Kilogram per Second (kg/s)
Mass Flow Rate Unit Conversion 
- **Measurement: Specific Energy** in Kilojoule per Kilogram (kJ/kg)
Specific Energy Unit Conversion 



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